

1. Solution:

In the lectures we have the formula

$$\tau = \frac{2\pi}{\sqrt{G(m_1 + m_2)}} a^{3/2}$$

but if $m_1 \gg m_2$, we can approximate

$$\tau \approx \frac{2\pi}{\sqrt{Gm_1}} a^{3/2}.$$

Now we use this formula for two different cases:

$$\tau_{EM} \approx \frac{2\pi}{\sqrt{Gm_E}} a_{EM}^{3/2} \quad \text{the Earth-Moon pair}$$

$$\tau_{ES} \approx \frac{2\pi}{\sqrt{Gm_S}} a_{ES}^{3/2} \quad \text{the Earth-Sun pair.}$$

Thus the masses are

$$m_E = \frac{4\pi^2}{G\tau_{EM}^2} a_{EM}^3$$

$$m_S = \frac{4\pi^2}{G\tau_{ES}^2} a_{ES}^3$$

and now it is easy to calculate the ratio of the masses

$$\frac{m_E}{m_S} = \left(\frac{\tau_{ES}}{\tau_{EM}} \right)^2 \left(\frac{a_{EM}}{a_{ES}} \right)^3.$$

Using given values results to

$$\frac{m_E}{m_S} \approx 2.97 * 10^{-6}$$

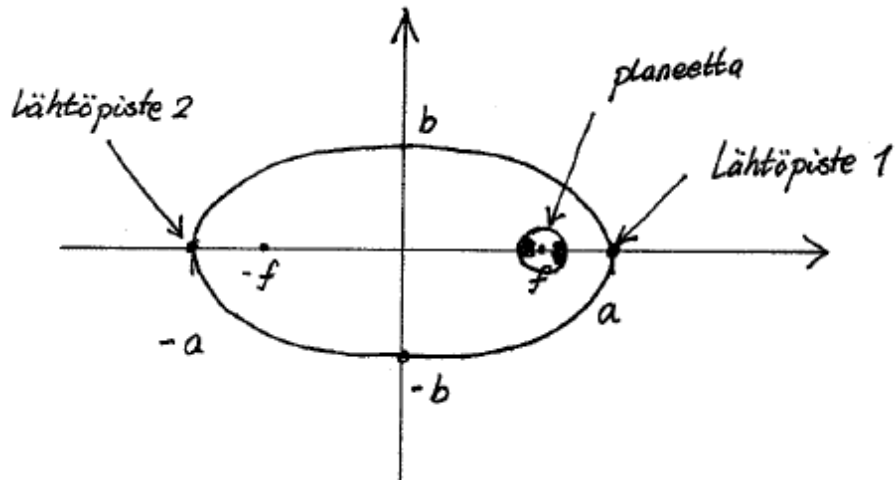
or in a similar way

$$\frac{m_S}{m_E} \approx 337000 \quad (\text{real value} \approx 333000).$$

2. Solution:

The satellite either starts from the point 1 or 2 ($\pm a$) and it has gravitational potential

$$V = -\frac{k}{m} = -\frac{GMm}{r}.$$



We know the eccentricity of the orbit

$$\epsilon = \sqrt{1 + \frac{2El^2}{m'k^2}}$$

where m' is the reduced mass

$$m' = \frac{mM}{m+M} = \frac{m}{\frac{m}{M} + 1} \approx m, \quad m \ll M$$

and l is the angular momentum that is conserved, meaning that

$$l = l_{\text{beginning}} = mrv.$$

Now the eccentricity has a form

$$\begin{aligned} \epsilon &= \left[1 + \frac{2\left(\frac{1}{2}mv^2 - \frac{GMm}{r}\right)(mrv)^2}{m(GMm)^2} \right]^{1/2} \\ &= \left[1 - 2\frac{rv^2}{GM} + \frac{r^2v^4}{G^2M^2} \right]^{1/2} \\ &= \sqrt{\left(1 - \frac{rv^2}{GM}\right)^2} \\ &= \left| 1 - \frac{rv^2}{GM} \right|. \end{aligned}$$

We have different orbits depending on different eccentricities:

circle

$$\epsilon = 0 \Rightarrow 1 - \frac{rv^2}{GM} = 0 \Rightarrow v = \sqrt{\frac{GM}{r}},$$

parabola

$$\epsilon = 1 \Rightarrow 1 - \frac{rv^2}{GM} = \pm 1 \Rightarrow v = \sqrt{\frac{2GM}{r}} \quad \text{or} \quad v = 0,$$

and *hyperbola*

$$\begin{aligned}
 \epsilon &> 1 \\
 \Rightarrow 1 - \underbrace{\frac{rv^2}{GM}}_{\substack{>0 \\ <1}} &> 1 \quad \text{or} \quad 1 - \frac{rv^2}{GM} < -1 \\
 \Rightarrow \frac{rv^2}{GM} &> 2 \\
 \Rightarrow v &> \sqrt{\frac{2MG}{r}}.
 \end{aligned}$$

For an *elliptic orbit* the velocity has to be in the interval

$$0 < v < \sqrt{\frac{2MG}{r}}.$$

Because the case $v = \sqrt{\frac{GM}{R}}$ is the circle orbit, the starting point of the satellite is point 1, if $\sqrt{MG/r} < v < \sqrt{2MG/r}$, and point 2, if $0 < v < \sqrt{MG/r}$.

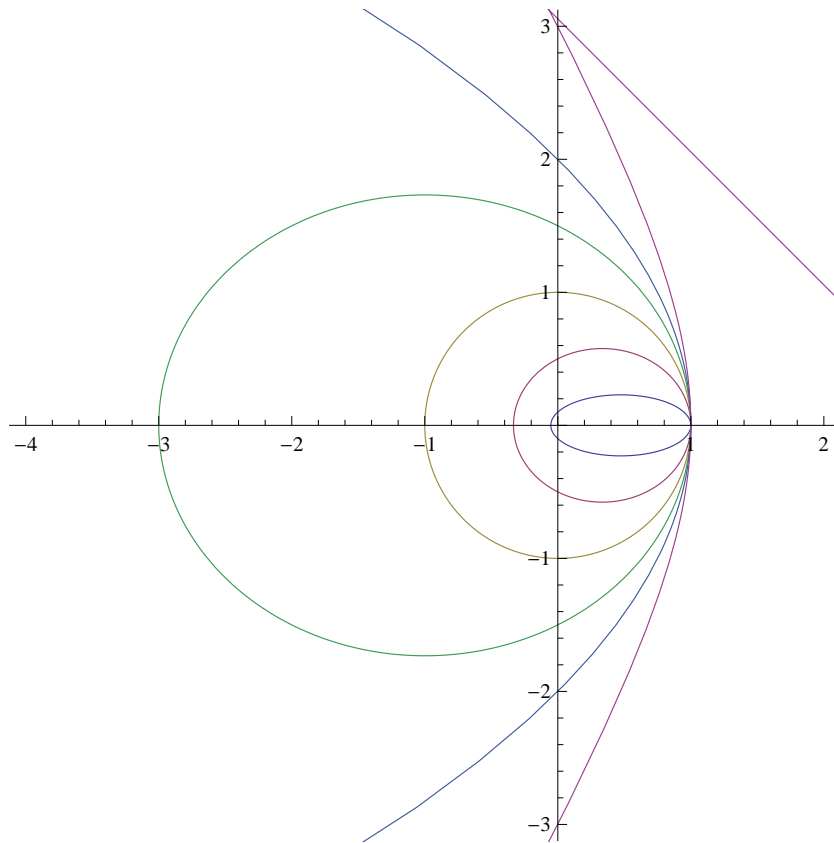


Figure 1: *Different orbits*

3. Solution:

We start from the formula in the lectures

$$\phi = \phi_0 \pm \frac{l}{\sqrt{2m}} \int_{r_0}^r \frac{dr}{r^2 \sqrt{E - V_{\text{eff}}}}. \quad (1)$$

Now we are interested in an angular difference $\Delta\phi$ that is the angle of the two sequential extreme values of r . Let's substitute $r = r_{\text{max}}$ and $r_0 = r_{\text{min}}$:

$$\Delta\phi = \phi - \phi_0 = \frac{l}{\sqrt{2m}} \int_{r_{\text{min}}}^{r_{\text{max}}} \frac{dr}{r^2 \sqrt{E - V_{\text{eff}}}}. \quad (2)$$

The next problem is to find the extrema values of r . Naturally these are points, where $E - V_{\text{eff}}$ vanish (see the lectures). Let's assume that r is close to r_0 . In this environment we can expand the potential V_{eff} as Taylor series up to second-order

$$V_{\text{eff}}(r) \approx V_{\text{eff}}(r_0) + \underbrace{V'_{\text{eff}}(r_0)}_{=0} (r - r_0) + \frac{1}{2} V''_{\text{eff}}(r_0) (r - r_0)^2$$

and thus

$$E - V_{\text{eff}}(r) \approx E - \underbrace{V_{\text{eff}}(r_0)}_{\equiv E_0} - \frac{1}{2} V''_{\text{eff}}(r_0) (r - r_0)^2.$$

Now it is easy to solve

$$E - V_{\text{eff}}(r_{\text{min,max}}) = 0$$

meaning

$$r_{\text{min,max}} = r_0 \pm \sqrt{\frac{2(E - E_0)}{V''_{\text{eff}}(r_0)}}.$$

So our integration limits are for now checked. After this let's look the integrand. Because

$$r \sqrt{E - E_0} = r_0 \sqrt{E - E_0} + \underbrace{(r - r_0)}_{\text{small}} \underbrace{\sqrt{E - E_0}}_{\text{small}} \approx r_0 \sqrt{E - E_0}$$

we can replace r with r_0 . In the above we approximate that small times small is zero. Our integral gets easier with changing variables:

$$r - r_0 = s \sqrt{\frac{2(E - E_0)}{V''_{\text{eff}}(r_0)}}.$$

Now we have to check again the limits of the integral: when $r = r_{\text{max}}$, it is clearly $s = 1$ and when $r = r_{\text{min}}$ implies $s = -1$. Furthermore we have

$$dr = \sqrt{\frac{2(E - E_0)}{V''_{\text{eff}}(r_0)}} ds$$

and

$$E - V_{\text{eff}}(r) = (E - E_0)(1 - s^2).$$

After all this we get

$$\begin{aligned} \Delta\phi &= \frac{l}{\sqrt{2m}} \int_{-1}^1 \frac{1}{r_0^2} \frac{1}{\sqrt{E - E_0}} \frac{1}{\sqrt{1 - s^2}} \sqrt{\frac{2(E - E_0)}{V_{\text{eff}}''(r_0)}} ds \\ &= \frac{l}{r_0^2 \sqrt{mV_{\text{eff}}''(r_0)}} \int_{-1}^1 \frac{ds}{\sqrt{1 - s^2}} \\ &= \frac{l\pi}{r_0^2 \sqrt{mV_{\text{eff}}''(r_0)}} \end{aligned}$$

where the last integral can be solved with e.g. change of variables $s = \sin \alpha$. Now we fix the potential and start to study a potential that has a form $V(r) = ar^{n-1}$. This means that

$$V_{\text{eff}}(r) = \frac{l^2}{2mr^2} + ar^{n+1}.$$

We can also calculate minimum at the point r_0 where $V_{\text{eff}}'(r_0) = 0$ meaning

$$V_{\text{eff}}'(r) = -2\frac{l^2}{2mr^3} + a(n+1)r^n.$$

The most useful thing is to solve the constant a

$$a = \frac{l^2}{(n+1)mr_0^{n+3}}.$$

The only thing we still need is $V_{\text{eff}}''(r_0)$:

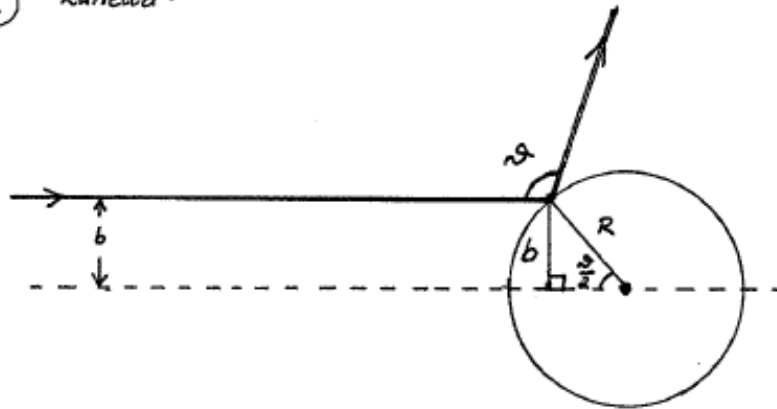
$$\begin{aligned} V_{\text{eff}}''(r_0) &= \frac{3l^2}{mr_0^4} + a(n+1)nr_0^{n-1} \\ &= \frac{3l^2}{mr_0^4} + \frac{l^2}{(n+1)mr_0^{n+3}}(n+1)nr_0^{n-1} \\ &= \frac{3l^2}{mr_0^4} + \frac{l^2n}{mr_0^4} \\ &= \frac{l^2}{mr_0^4}(n+3). \end{aligned}$$

By substituting our result into the equation of $\Delta\phi$ we get

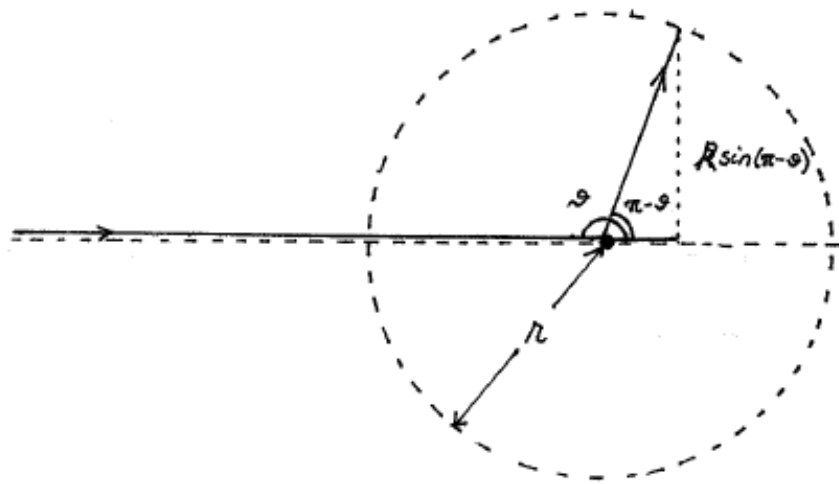
$$\Delta\phi = \frac{\pi}{\sqrt{3+n}}.$$

4. Solution:

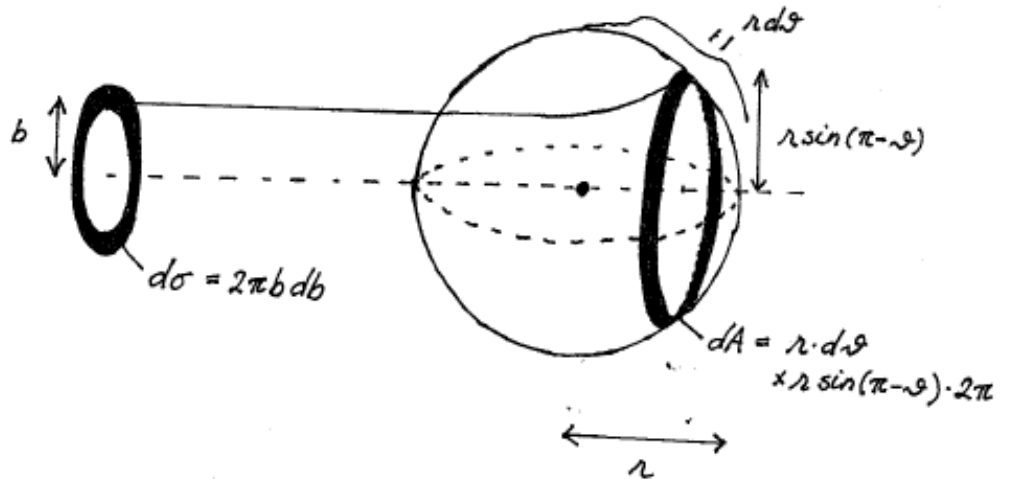
3. Läheltä:



Kaukaa:



3D:



From the picture we notice that using trigonometric identities we get

$$\sin \frac{\theta}{2} = \frac{b}{R}$$

and differentiating this leads us to (*)

$$\begin{aligned} \frac{1}{2} \cos \frac{\theta}{2} d\theta &= \frac{db}{R} \\ \Rightarrow d\theta &= \frac{2db}{R \cos \frac{\theta}{2}}. \end{aligned}$$

From the interval $[b, b+db]$, meaning that particles coming from the area $2\pi b db$ scatter to an angle $[\theta, \theta + d\theta]$ and thus they are leaving from the scattering center through a circle that has an area

$$dA = r d\theta * 2\pi r \sin(\pi - \theta).$$

So the solid angle where the particles are scattering is

$$\begin{aligned} d\Omega &= \frac{dA}{r^2}, \quad \text{the definition of the solid angle} \\ &= 2\pi \sin(\pi - \theta) d\theta, \quad \sin(\pi - \alpha) = \alpha \quad \forall \alpha \\ &= 2\pi \sin \theta d\theta, \quad \text{we know } d\theta \\ &= 2\pi \sin \theta \frac{2db}{R \cos \frac{\theta}{2}}, \quad \sin \alpha = 2 \sin \frac{\alpha}{2} \cos \frac{\alpha}{2} \\ &= 8\pi \sin \frac{\theta}{2} \frac{db}{R}. \end{aligned}$$

Furthermore, $d\sigma = 2\pi b db = 2\pi R \sin(\theta/2) db$. Now the differential cross section is

$$\frac{d\sigma}{d\Omega} = \frac{2\pi R \sin(\theta/2) db}{8\pi \sin \frac{\theta}{2} \frac{db}{R}} = \frac{R^2}{4}.$$

and the total cross section is

$$\sigma = \int_{\Omega} \frac{d\sigma}{d\Omega} d\Omega = \int_0^{2\pi} \int_0^{\pi} \frac{R^2}{4} \sin \theta d\theta d\phi = \frac{R^2}{4} * 4\pi = \pi R^2.$$

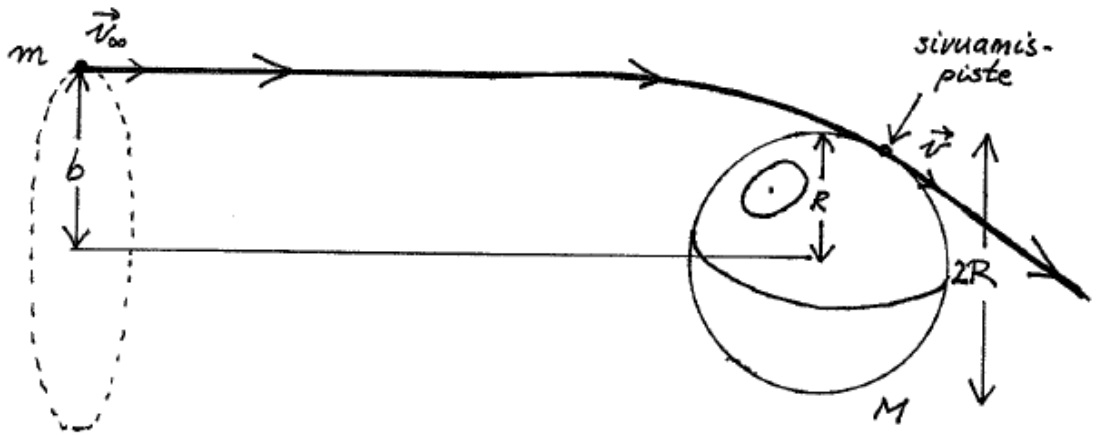
The second way of calculating the differential cross section is to notice from the begin (*)

$$\frac{db}{d\theta} = \frac{1}{2} R \cos \frac{\theta}{2}$$

and use the formula in the lectures

$$\frac{d\sigma}{d\Omega} = \frac{b}{\sin \theta} \left| \frac{db}{d\theta} \right| = \frac{1}{2} R^2 \frac{\cos \frac{\theta}{2} \sin \frac{\theta}{2}}{\sin \theta} = \frac{R^2}{4}.$$

5. Solution:



The collision cross section is the area that a particle has to go through to collide with the scatterer. Here our scatterer is a sphere. Thus the area through which a particle collides is a disk meaning that it can be presented as

$$\sigma_{\text{col}} = \pi b^2.$$

So our only problem is to solve what is b . Without gravity it is clear that b would be R but now we have to notice gravity that pulls our particle towards our sphere. Our requirement is that a particle leaving with the impact parameter b will touch the sphere. Far, far a way from the sphere gravitational potential is zero (approximately) and the particle has energy

$$E = \frac{1}{2}mv_\infty^2.$$

The angular momentum is

$$L = mv_\infty b.$$

On the other hand in the point of the contact the gravitational potential is

$$V = -\frac{GmM}{R}$$

and the energy is thus

$$E = \frac{1}{2}mv^2 - \frac{GmM}{R}.$$

In the point of contact the angular momentum is

$$L = mvR.$$

Because the angular momentum is a conserved, we have a relation

$$mv_{\infty}b = mvR \Rightarrow v = \frac{b}{R}v_{\infty}.$$

Also particle's energy is conserved quantity:

$$\begin{aligned} \frac{1}{2}mv_{\infty}^2 &= \frac{1}{2}mv^2 - \frac{GmM}{R} \\ \Leftrightarrow \\ b^2 &= R^2 + \frac{2GMR}{v_{\infty}^2}. \end{aligned}$$

So the cross section is

$$\sigma_{\text{col}} = \pi b^2 = \pi R^2 + \frac{2\pi GMR}{v_{\infty}^2}.$$

This can be done also with the lecture notes. The hint tells that $f - a = R \Rightarrow f = R + a$. From the lectures one gets

$$\begin{aligned} b^2 &= f^2 - a^2 \\ &= (R + a)^2 - a^2 \\ &= R^2 - 2Ra \\ &= R^2 + \frac{R|k|}{E} \\ &= R^2 + \frac{2R|k|}{mv_{\infty}^2} \\ &= R^2 + \frac{2RGM}{v_{\infty}^2}, \end{aligned}$$

where one uses from the lectures

$$a = \frac{|k|}{2E}, \quad E = \frac{1}{2}mv_{\infty}^2 \quad \text{and} \quad |k| = GmM.$$